

HW Set #7 - C. Greene's solutions

5. 11.6 and 11.22

Problem 11.6 (a) Let K' be the instantaneous rest frame of the rocket, and K be the earth's rest frame.

To relate the acceleration in frames K and K' , consider a small velocity u' in K' changing at the rate

$$\frac{du'}{dt'} = g$$

The velocity of the ship, according to an observer in the earth frame K , is

$$u = \frac{v + u'}{1 + \frac{vu'}{c^2}}$$

$$\Rightarrow du = \frac{du'}{1 + \frac{vu'}{c^2}} - \frac{(v + u') \frac{v}{c^2}}{\left(1 + \frac{vu'}{c^2}\right)^2} du'$$

$$= \frac{\left(1 + \frac{vu'}{c^2}\right) - \frac{v^2}{c^2} - \frac{u'v}{c^2}}{\left(1 + \frac{vu'}{c^2}\right)^2} du'$$

$$\text{or } du = \frac{1 - \frac{v^2}{c^2}}{\left(1 + \frac{vu'}{c^2}\right)^2}$$

or see Jackson #11.5

$$\Rightarrow \frac{du}{dt} = \frac{\left(1 - \frac{v^2}{c^2}\right) du'}{\gamma(dt' + \frac{v}{c^2} dx') \left(1 + \frac{vu'}{c^2}\right)^2} = \frac{\left(1 - \frac{v^2}{c^2}\right)^{3/2}}{\left(1 + \frac{vu'}{c^2}\right)^3} \frac{du'}{dt'} = \frac{du}{dt}$$

$$(11.6) (a) (p.2) \quad \text{Or at } u' \rightarrow 0, \quad \frac{du}{dt} = \frac{g}{\gamma^3}$$

acceleration as seen by an observer on earth
and note that $v = u$ since $u' \rightarrow 0$

$$\Rightarrow \frac{du}{dt} = \frac{dv}{dt} \Rightarrow \int_0^v \frac{dv}{\left(1 - \frac{v^2}{c^2}\right)^{3/2}} = gt$$

$$\Rightarrow \gamma v = gt \quad \text{or} \quad v = \frac{gt}{\sqrt{1 + \frac{g^2 t^2}{c^2}}}$$

$$\text{and } \gamma = \frac{gt}{v} = \sqrt{1 + \frac{g^2 t^2}{c^2}}$$

To relate the clocks in K, K' , we

$$dt' = \frac{dt}{\gamma} = \frac{dt}{\sqrt{1 + \frac{g^2 t^2}{c^2}}} \Rightarrow t' = \int_0^t \frac{dt}{\sqrt{1 + \frac{g^2 t^2}{c^2}}}$$

$$\Rightarrow t' = \frac{c}{g} \sinh^{-1}\left(\frac{gt}{c}\right) \quad \text{or} \quad t = \frac{c}{g} \sinh\left(\frac{gt'}{c}\right)$$

$$\Rightarrow \text{using } g = 9.8 \frac{\text{m}}{\text{s}^2}, \quad \left. \begin{array}{l} t' = 5 \text{ years} \\ t = 83.8 \text{ years} \end{array} \right\} \begin{array}{l} \text{first} \\ \text{stage} \end{array}$$

second stage is the same as the first stage,
by symmetry, $\Rightarrow t_{\text{trip}} = 4 \times 83.8 = 335 \text{ years}$

\Rightarrow The date when the rocket returns to earth
will be $2100 + 335 = \boxed{2435 \text{ AD}}$

(11.6)(b) During the 1st stage,

$$x = \int_0^t v dt = \int_0^t \frac{g t' dt'}{\sqrt{1 + \frac{g^2 t'^2}{c^2}}}$$

$$\text{or } x = g \frac{c^2}{g^2} \left[\left(1 + \frac{g^2 t^2}{c^2} \right)^{1/2} - 1 \right]$$

$$\Rightarrow \frac{x}{c} = \frac{c}{g} \left[\left(1 + \frac{g^2 t^2}{c^2} \right)^{1/2} - 1 \right]$$

$$= 82.8 \text{ light years}$$

And the same distance is covered in stage 2, so the maximum distance from earth is 166 light years

11.22 $E_0 =$ energy of the background radiation, at 3°K , i.e. $E_0 = k_B T$, $k_B = 8.6 \times 10^{-5} \text{eV/K}$

\Rightarrow Consider a photon E_0 colliding head-on with one of energy E_1 , in the "universe frame" K .



\Rightarrow In K , the total 4-momentum of the system looks like:

$$p^\alpha = \left(\frac{E_0 + E_1}{c}, \frac{E_1 - E_0}{c}, 0, 0 \right)$$

since the momentum of a photon with energy E is $p = E/c$.

Call K' the CM frame in which the total momentum is zero. In this frame the $e^+ - e^-$ pair produced will have zero momentum, which corresponds to the minimum energy for formation

$$\Rightarrow \text{In } K', \quad p'^\alpha = \left(\frac{E_{\text{cm}}}{c}, 0 \right)$$

Now $p^\alpha p_\alpha =$ Lorentz invariant

$$\Rightarrow E_{\text{cm}} = \sqrt{(E_0 + E_1)^2 - (E_1 - E_0)^2} = 2E_0 E_1$$

$$\text{At threshold, } E_{\text{cm}} = 2m_e c^2 = 2(5.11 \times 10^5 \text{eV})$$

$$(a) \Rightarrow \boxed{E_1 = \frac{(m_e c^2)^2}{E_0} = 1.0 \times 10^{15} \text{eV}} \text{ for } E_0 = (3\text{K}) k_B$$

$$(b) \quad E_{\text{inc}} = \frac{(5.11 \times 10^5 \text{eV})^2}{500 \text{eV}} = 5.2 \times 10^8 \text{eV}$$